

Jet Propulsion Laboratory
California Institute of Technology

Low Frequency Wire Grid Reflector for Space

T.B.H. Kuiper
Jet Propulsion Laboratory
California Institute of Technology

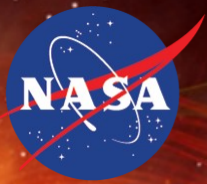
Science at Low Frequencies III

https://js.jpl.nasa.gov/JPLSpaceTemplates/JPL_PowerPointPresentationDownloadLinks.pdf

Pasadena, California

Originally contributed to the KISS Study Program

Planetary Magnetic Fields
Planetary Interiors and Habitability



Jet Propulsion Laboratory
California Institute of Technology

Low Frequency Wire Grid Reflector for Space

Objective:

Is a 1-km diameter reflector for operation at 0.1 to 3.75 MHz feasible given current launch capabilities?

Approach:

Design a minimum mass paraboloid wire mesh reflector anchored by small maneuverable satellites.

Wire Size and Spacing

Reflectivity of a sheet:

$$\Gamma(R_s) = \frac{R_s - R_0}{R_s + R_0}$$

For 95% and $R_0 = 377\Omega$, $R_s \leq 10\Omega$.

Resistance of a grid cell $L \times W$ in size:

$$R = R_s \frac{L}{W} \simeq R \leq 10\Omega$$

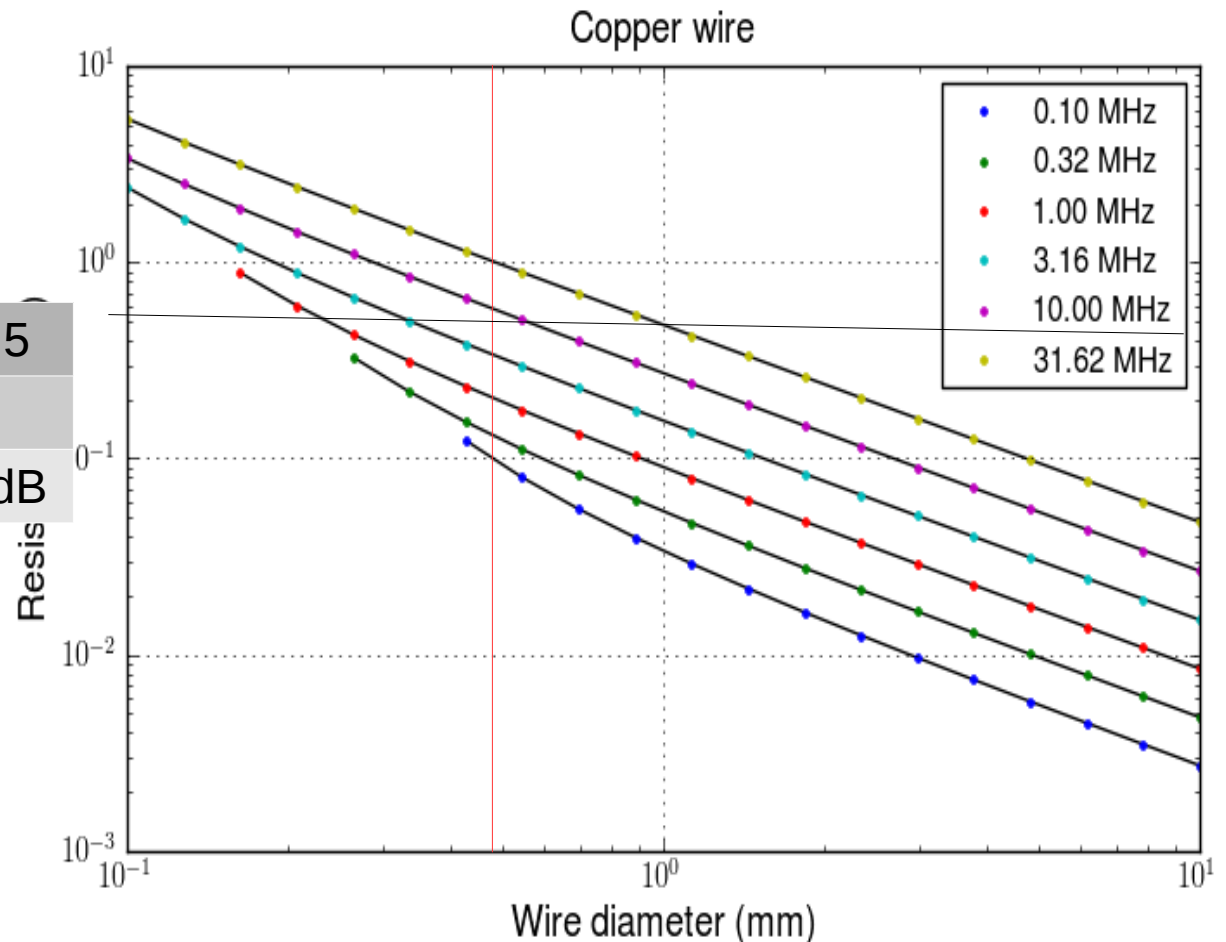
Resistance & Skin Depth:

$$R(d) = \frac{\rho}{\pi \delta(d - \delta)}$$

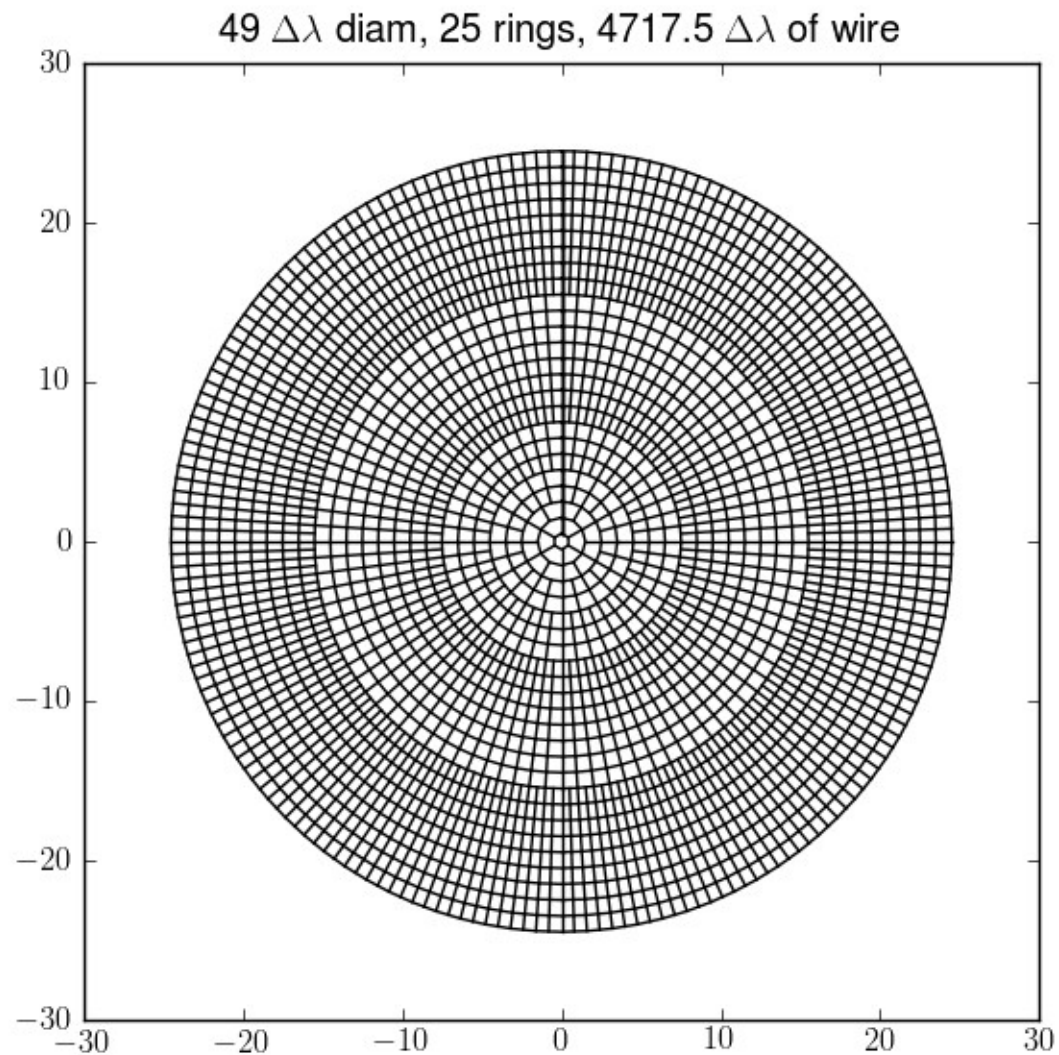
Cell size:

spacing	$\lambda / 2$	$\lambda / 4$	$\lambda / 5$
reflectivity	0.6	0.9	
directivity	31 dB	28 dB	26 dB

3.75 MHz (80 m wavelength)
requires a cell size of 20 m
which implies a wire **0.5 mm**
in diameter.



Wire Arrangement

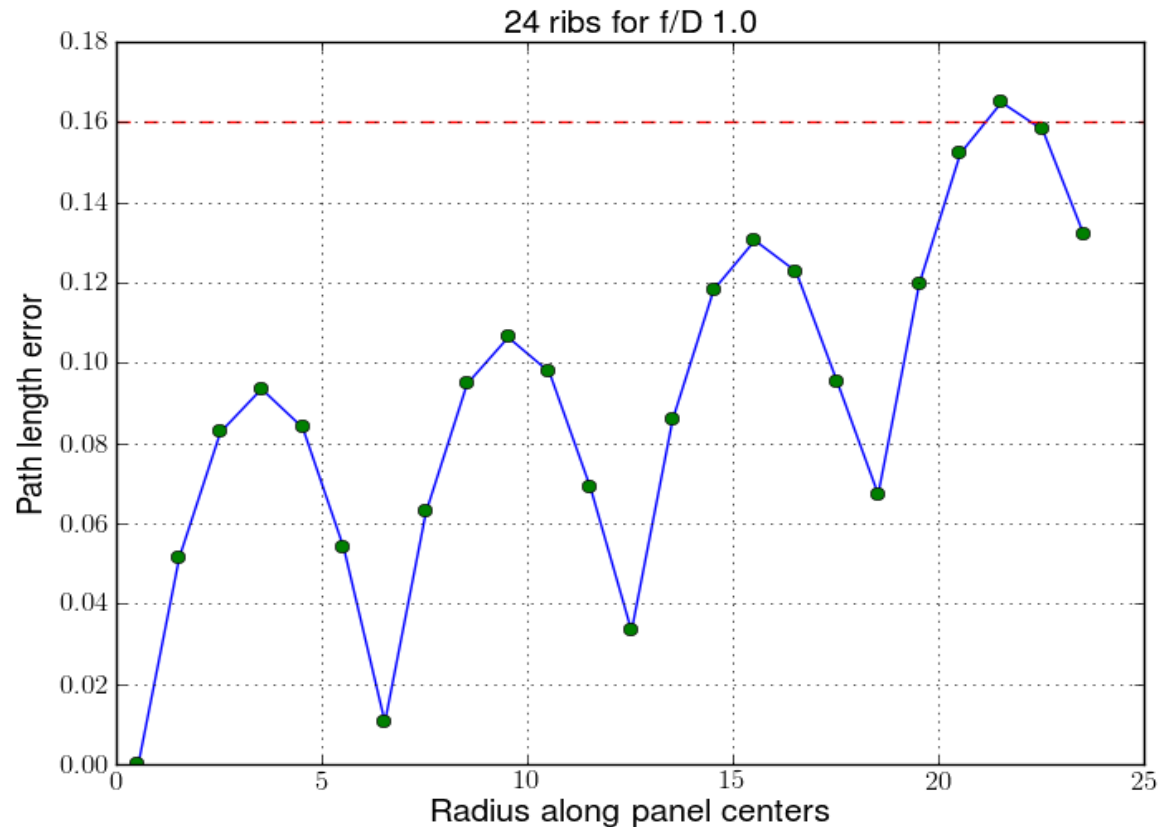


Anchoring

Radial

Azimuthal

f/D	0.25				0.6		1.0	
anchors	6	7	8	9	4	5	4	5
r.m.s. error	0.16	0.12	0.08	0.06	0.29	0.10	0.18	0.06
max. error	-0.30	-0.26	-0.16	-0.15	-0.56	-0.14	-0.35	-0.08
err. radius	6.5	12.5	2.5	8.5	18.5	2.5,4.5	18.5	var.



Mass of the Mesh Paraboloid

Mesh Mass

Wire length: $4717.5 \times 20 \text{ m} = 94.35 \text{ km}$

Cross-section area = $\pi \times 0.0025^2 \text{ m}^2 = 2 \times 10^{-7} \text{ m}^2$

Volume = 0.018 m^3

Density of copper = 8940 kg m^{-3}

Mass = **169.7 kg**

Anchor Mass

Number of anchor satellites = 96

Mass of a 2U Cubesat = 4.5 kg

Mass = **430 kg**

Launch Capacity to LEO

Light : up to 2000 kg

Medium: 2000-20,000 kg

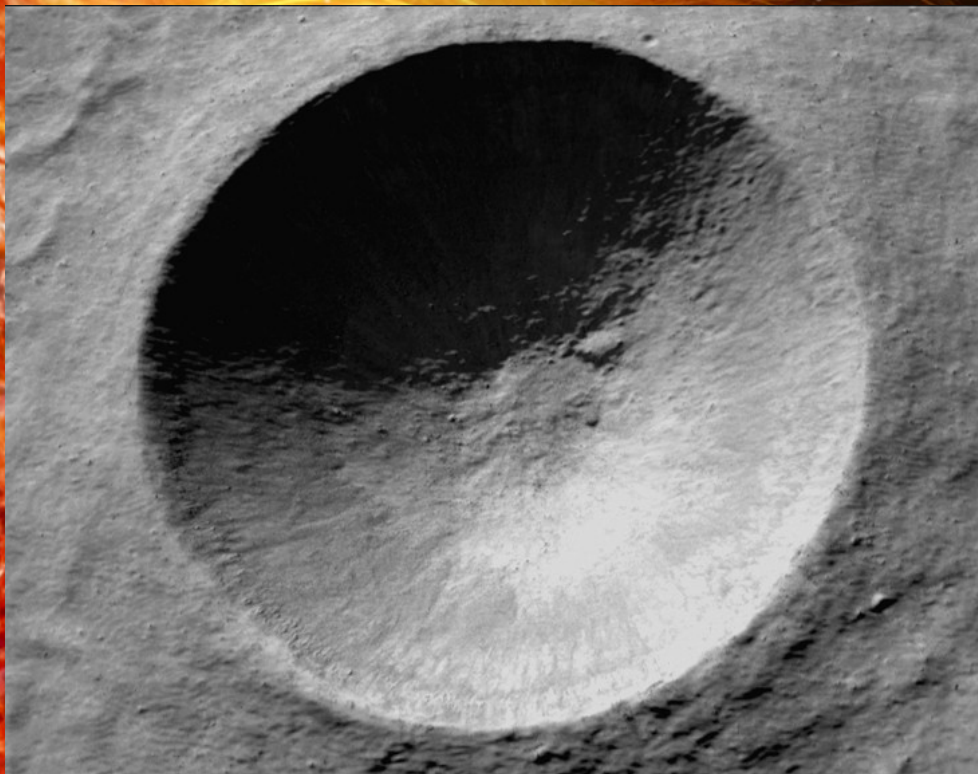
Heavy: 20,000 to 50,000 kg

Escape Velocity

To LEO: 9.3 km/s

To the Moon: 11.2 km/s

Another Approach to Anchoring



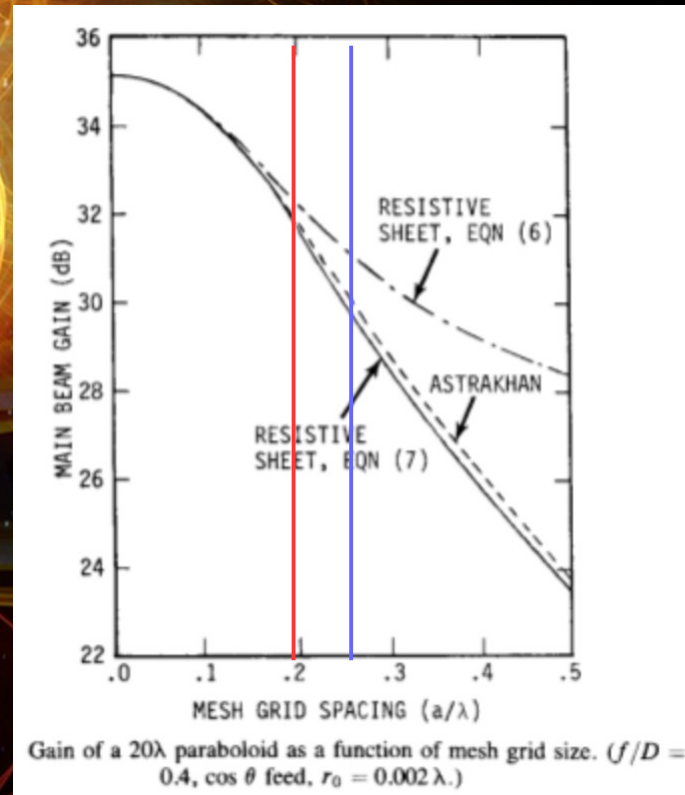
Linne Crater, 2.2 km diameter



Courtesy of Gregg Hallinan

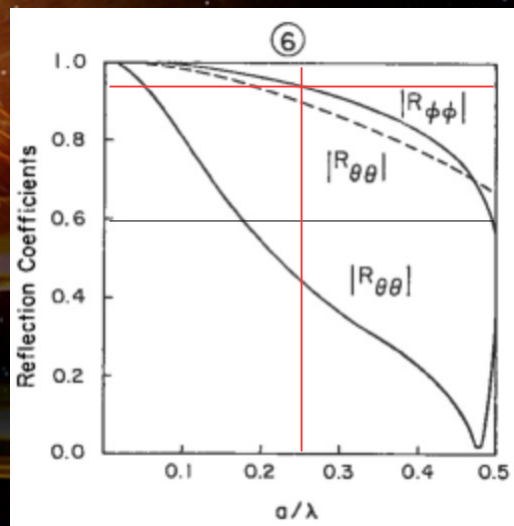


Backup: Paraboloid Mesh Gain



D. C. Jenn *et al.*, *IEEE Trans AP*, **37**, 1484 (1989)

Backup: Wire Mesh Reflectivity



D. A. Hill and J. R. Wait, *Can. J. Phys.*, **54**, 353 (1976).