CALIFORNIA INSTITUTE OF TECHNOLOGY JET PROPULSION LABORATORY

F STUDIES

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# Solar Sails and Strength of Materials

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## The Issue

- Solar sails have to transfer the distributed light pressure from the Sun back to a concentrated mass – the "spacecraft" or "observatory"
- For solar sails to effect "significant" propulsive capability, they need to have a "lightness number" λ ~ 1, that is, the entire "sailcraft" (sails plus support plus observatory) must be sufficiently low-mass to be pushed efficiently by light pressure
- This means that the strength of the supports for transferring / concentrating the pressure force to the acceleration of the spacecraft provide a limit to what can be accomplished with solar sail propulsion
- The question is how to deal with calculating this constraint in a fashion that illuminates the necessary trades

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## A Gedanken Experiment Approach

- Consider an ideal sailcraft
  - Perfectly reflecting sails
  - » "Point mass" observatory
  - Connected to the sail supports at the center of symmetry of the sail
  - Sail is taken as "flat" to first order
  - > Sail segments are triangular

#### Consider an ideal trajectory

- Sail normal pointed radially toward the Sun
- > Angular momentum is conserved

#### Consider an ideal configuration

- Sails supported by simple hollow spars
- Spars transfer the solar radiation pressure to the central observatory mass



## **Example of Stress Analysis**

- Each spar individually experience a torque about the entral mass support
- Detailed analysis is a series of papers by Greschik and Mikulas (2002) and others



Fig. 3 Parameters for 400-m stripped sails at 1 AU: payload,  $m_{\rm pl} = 320$  kg; sail film,  $\sigma_{\rm skin} = 6895$  Pa = 1 psi;  $t = 2.5 \ \mu$ m; sailcraft,  $m_{\rm tot} \sim 1200-1350$  kg;  $\bar{\rho} \sim 5.4-6.3$  g/m<sup>2</sup>; and performance,  $a_c \sim 0.9-1$  mm/s<sup>2</sup>.

#### 4-spar example

Light pressure is "down" from "top"

Gravity is up – so concentrated mass of the observatory pulls center-point up

Provides a potential mechanical failure point where spars are cantelivered

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## **Question Is What Materials Can Support**

- As the observatory mass increases, the required sail size and spar lengths increase
- If the spars cannot support the radiation pressure, then they will fail like an umbrella in a strong wind





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