

Ultralight photovoltaic power and fuel tanks

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- Ultralight modular integrated photovoltaics mass and size
 - Module level description
 - Ultralight photovoltaic specific power on Mars
- Fuel and tank mass and size

Large Scale Space Solar Power: Specific Power

SPS	USEF	JAXA	ESA	Alpha	Caltech
W/kg	41	98	132	33	2000- 6000
Max size for deployment	100 m x 95 m	3.5 km	15 km	6 km	60 m x 60 m

Current State of Art

- Off-the-shelf PV and power components

Watts on ground/kg in space

Ultralight Approach

- Innovate to develop science and technology for a lightweight, high-performance modular system; assume *current* launch costs

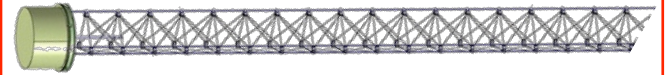
Module Design Summary

60 m x 60 m square architecture.
Designed for tile rotation $<1^\circ$ under SRP.

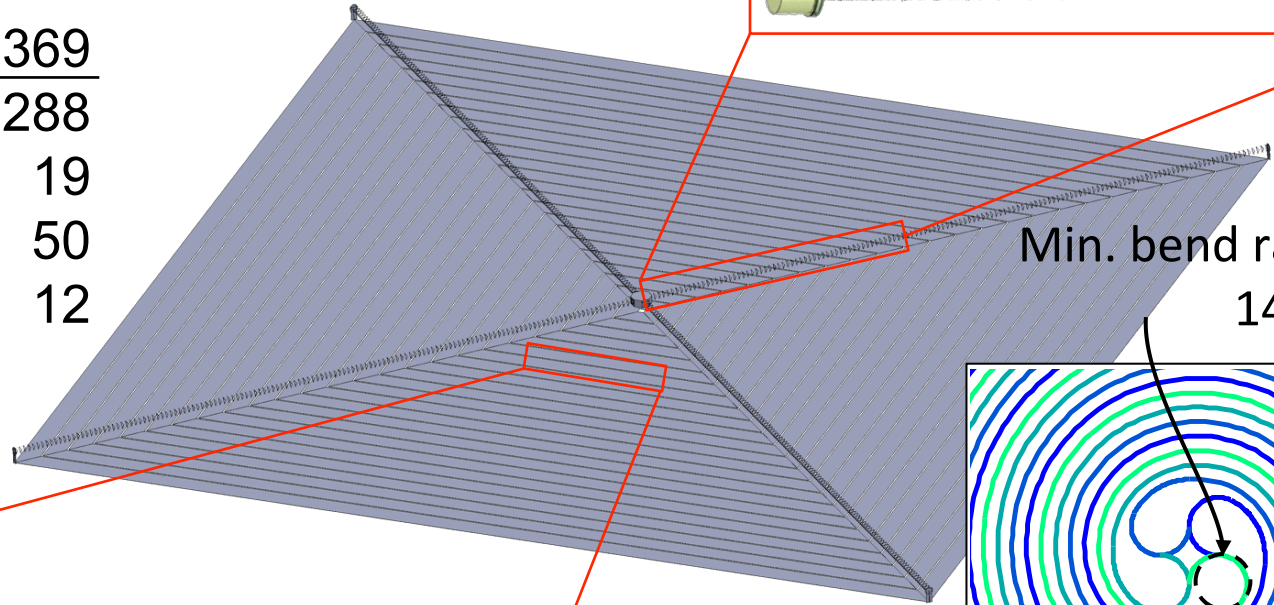
Mass (kg)	369
Tiles (@80 g/m ²)	288
Strip structure	19
Hub	50
Booms	12

Booms:

4 × coilaible CFRP trusses
42 m long, ~1.5 kg each, ATK



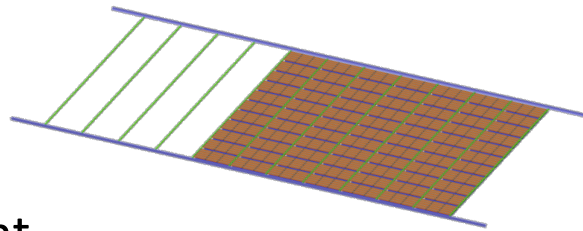
Min. bend radius
14 mm



Strips:

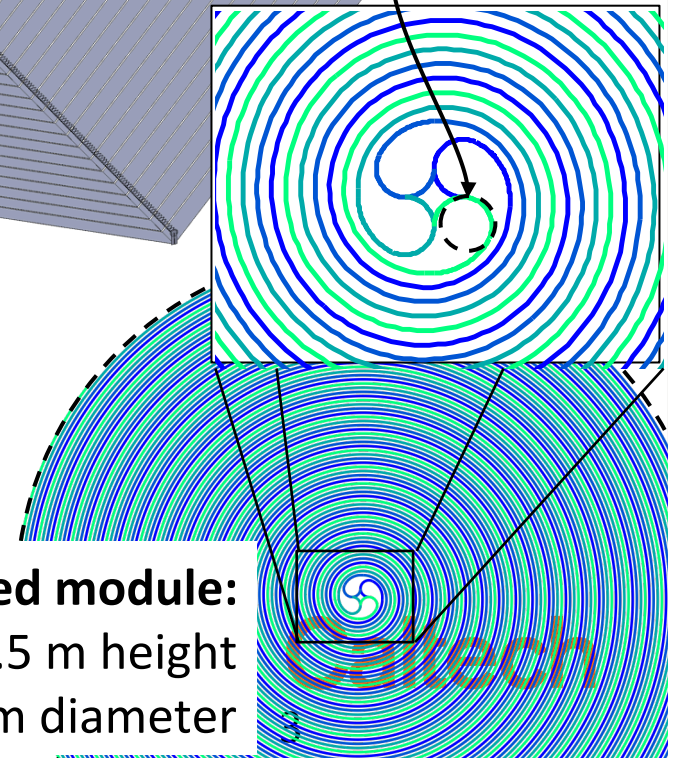
1.5 m wide
20 per quadrant
CFRP TRAC longerons (2 x 64 μm thick tape
springs, max stowed strain $< 0.5\%$)

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Stowed module:

1.5 m height
0.926 m diameter



Module Power on Mars

Space: 4.7 MW at AM0 (1330 W/m²)

Daily average on Mars, no dust: = 529 kW
(147 W/m² , from Dave)

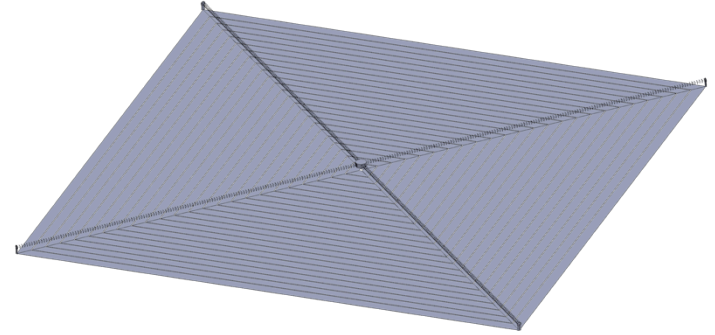
Daily average on Mars, dust storm: 118 kW =
(33 W/m² , from Dave)

Module weight = 370 kg

Specific power on Mars:

$529/370 = 1.4 \text{ kW/kg}$ (no dust)

$118/370 = 318 \text{ W/kg}$



Weight of fuel tanks

- Data derived from rocket motor tank specs
- Tank wall thickness:
 - Falcon 9 Heavy: 4.7mm (reusability)
 - Space Shuttle external tank and Atlas rocket ('old school'): 2.5 to 10 mm
 - Centaur upper stage: 0.36 to 0.41 mm
 - Saturn V first stage: 4.32 to 6.45 mm
- Tank volume and size:
 - Saturn V:
 - 43 m tall and 10 m diameter
 - 770,000 liters of kerosene and 1.2 million liters of liquid oxygen
- Tank and fuel weight (Saturn V):
 - $(314 \text{ m}^2 \times 0.003 \text{ m} = 0.94 \text{ m}^3) \times 2830 \text{ kg/m}^3 = 2600 \text{ kg}$ 2.6 metric tons
 - 770,000 liters of kerosene and 1.2 million liters of liquid oxygen: Assume it's all oxygen $(1.97 \text{ million liters of liquid oxygen}) \times (1.14 \text{ kg/l}) = 2.25 \text{ million kg} = 2,250 \text{ metric tons of liquid oxygen}$