

Small Satellite Laser Comm Pointing

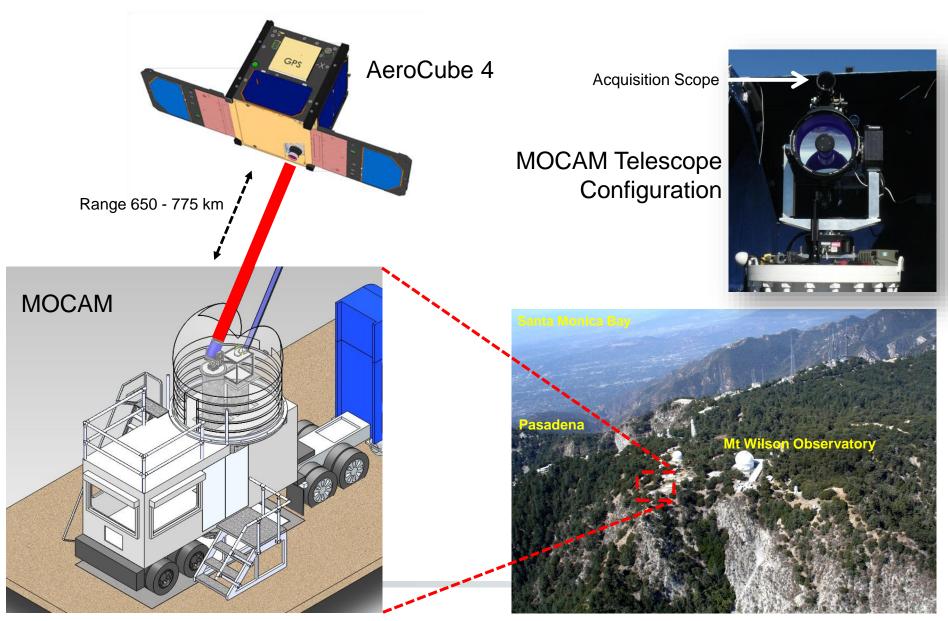
Darren Rowen

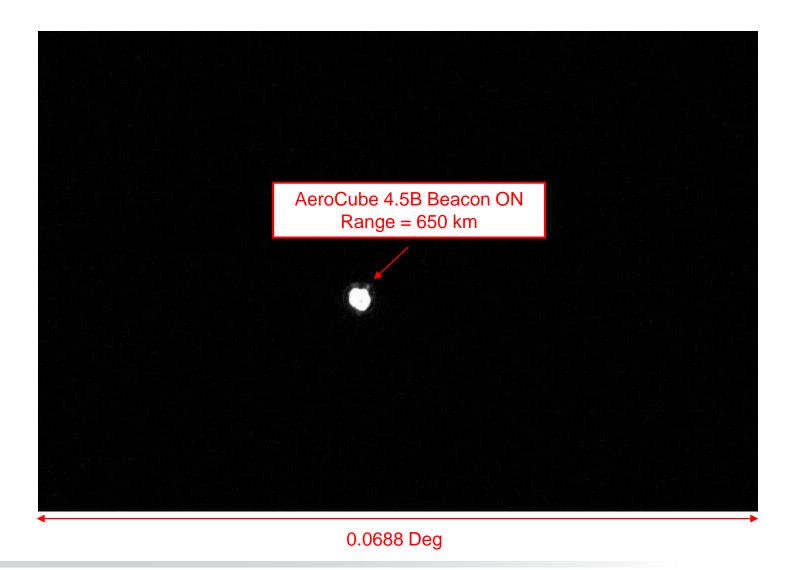
Microsatellite Systems Dept. July 11, 2016

Agenda

- Optical Ground Station Tracking Demo of Cubesat Laser
- OCSD-B/C Design & Configuration
- OCSD-A Star Tracker On-Orbit Performance
- CubeSat Star Tracker Industry Survey



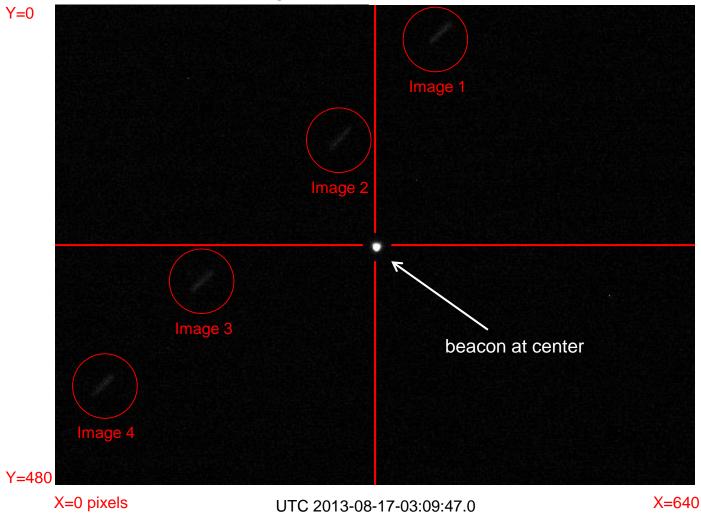




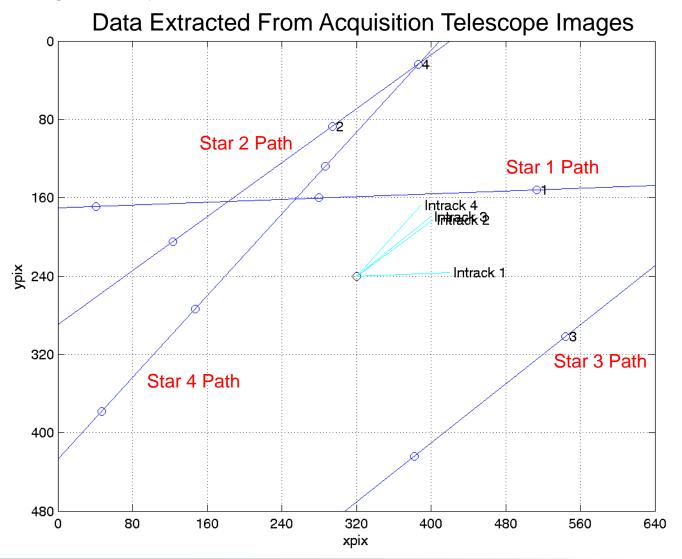


- Star Matching to Verify Ephemeris

Stacked Images From Acquisition Telescope



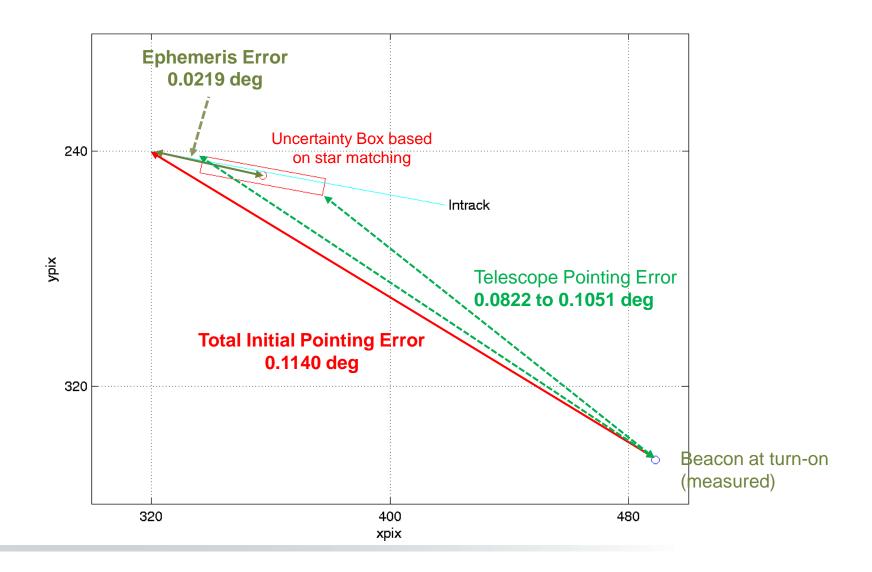
- Star Matching to Verify Ephemeris



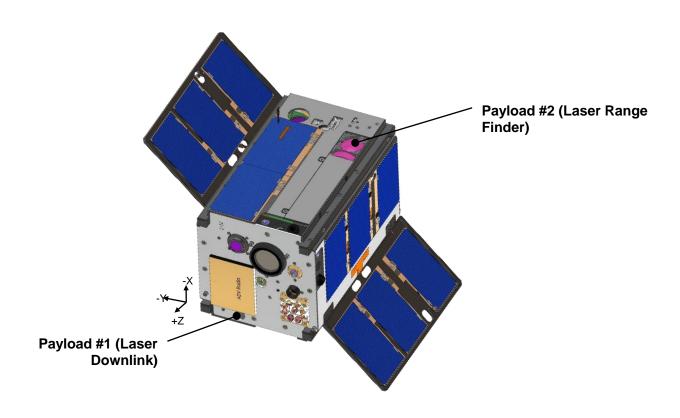




- Open Loop Acquisition Pointing Error Breakdown



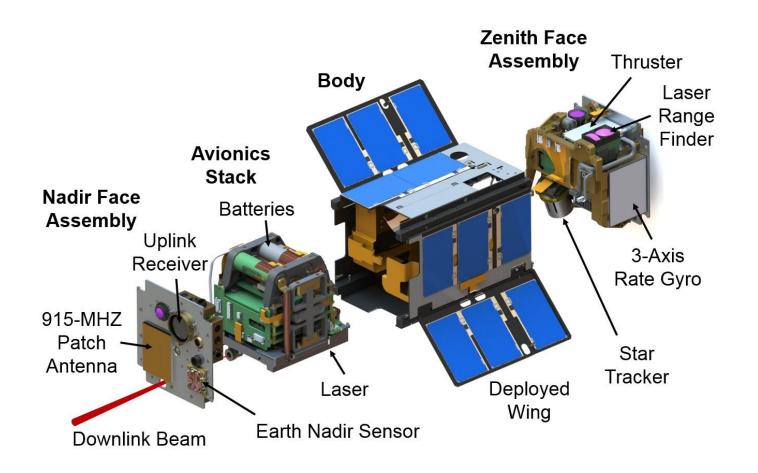
OCSD B/C: Vehicle Configuration



Key exterior features

OCSD B/C Layout

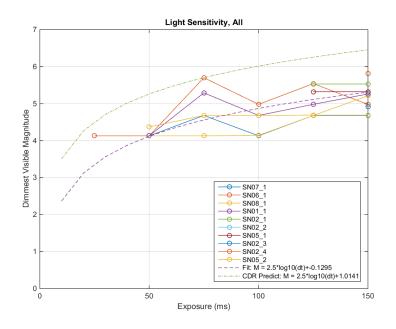
Exploded view

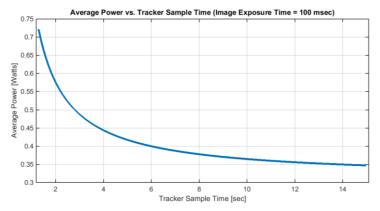




OCSD-B/C Star Camera Acceptance Test (Ground)

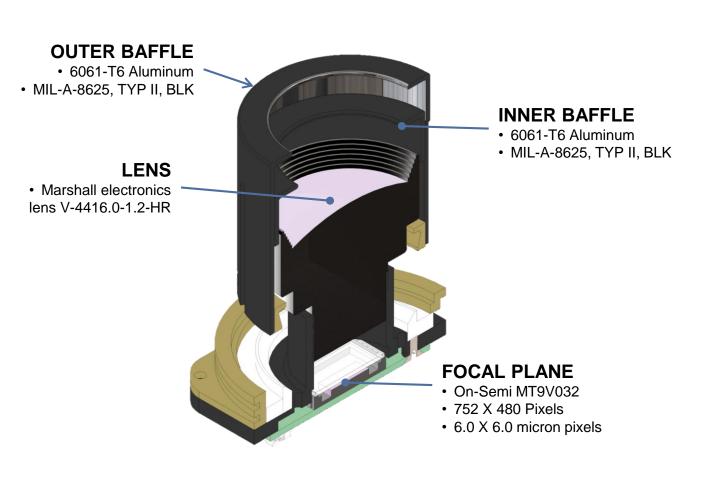
- OCSD B/C Star Tracker Units have been tested for light sensitivity, power consumption via dark sky testing
- Post-vibe verified functionality & focus in the lab, collected dark images for hot pixel mapping

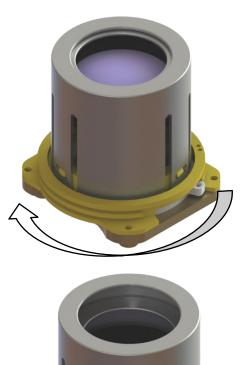






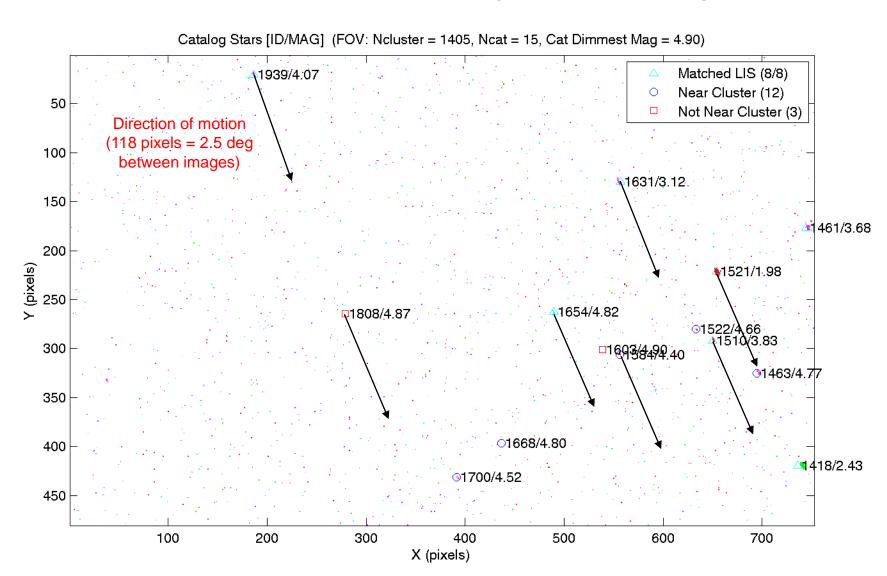
Star Tracker Design & On-orbit Results (OCSD-A)





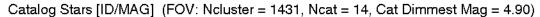


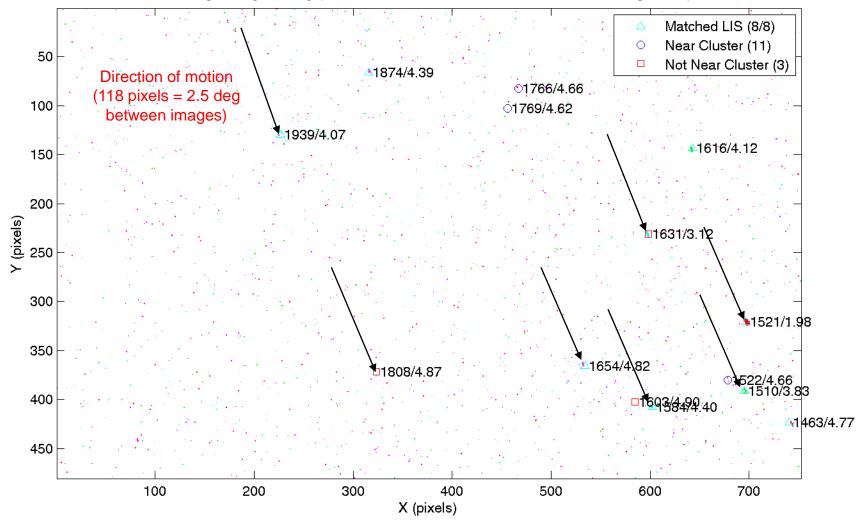
Post-Processed Star Tracker Image 1 (0.97 deg/s)





Post-Processed Star Tracker Image 2 (0.97 deg/s)





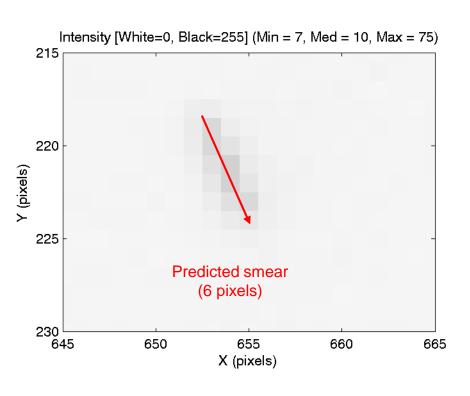


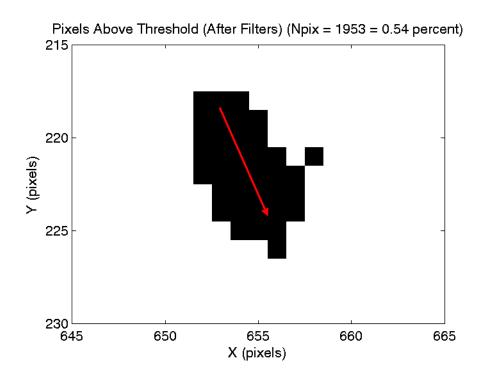
Star Smear Detail

Brightest Star (Mag 1.98)

Inverted Image

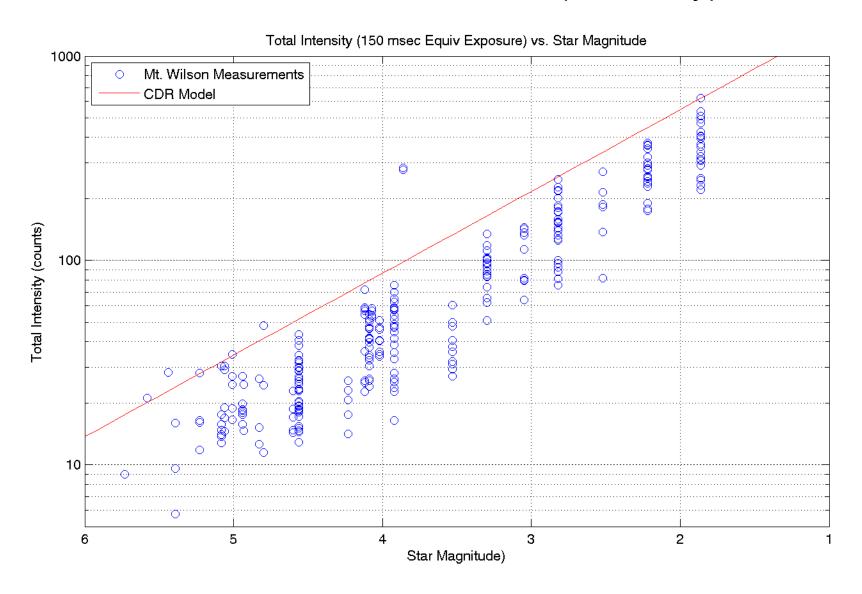
Pixels above threshold





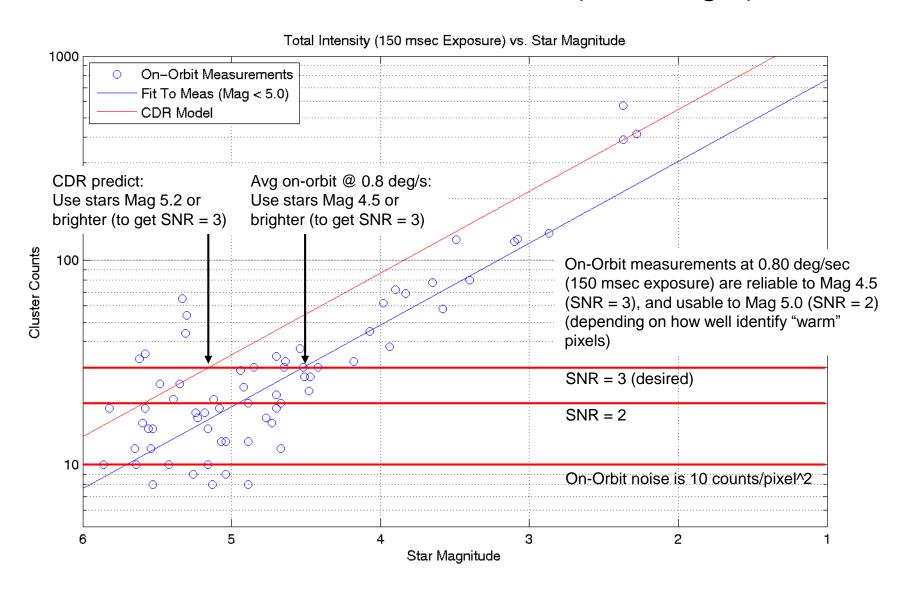


Star Tracker Ground Measurements (Dark Sky)



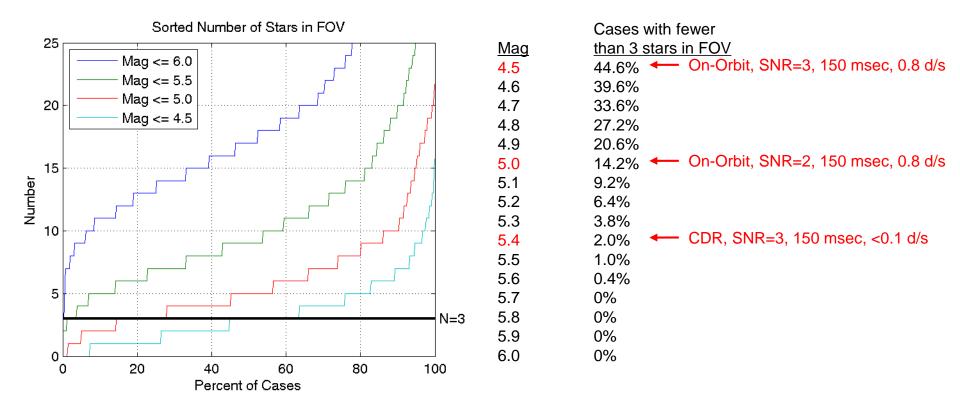


Star Tracker On-Orbit Measurements (0.80 deg/s)



Star Tracker Solution Probability

 On-orbit sensitivity data supports analytical studies to determine the probability of obtaining a valid solution at maximum mission slew rates

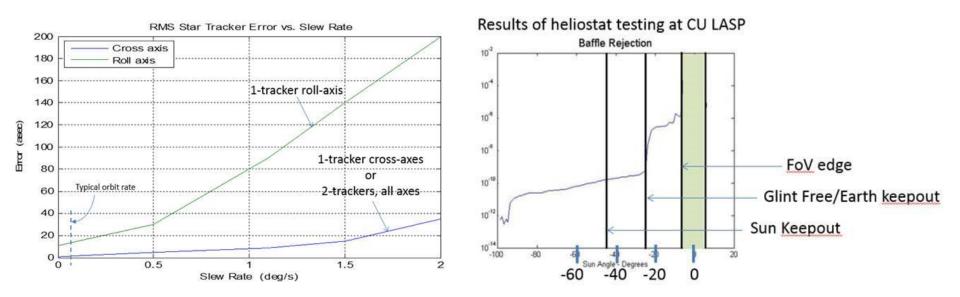


Monte Carlo 500 random attitudes



Blue Canyon Technologies XACT

- MinXSS CubeSat First to fly XACT system
 - 1-sigma cross-axis pointing error better than 8 arcseconds
 - Max tested on-orbit slew rates < 0.15 deg/s
 - Ground test data (below) suggests the following performance for higher slew rates. Note ground test data demonstrates knowledge accuracy of the star tracker but unable to demonstrate closed loop pointing accuracy due to limitations of ground testing.



Data provide by and presented with permission form Blue Canyon Technologies



CubeSat Star Tracker Performance & SWAP

- Star Tracker Performance Continues to Improve and Size Continues to Shrink
- Note: table shows accuracy at low slew rate, accuracies at higher slew rates are not well characterized for most star trackers

NanoSatellite StarTracker	Attitude accuracy (pitch/roll)	Size/Weight	Max Slew rate (availability >80% for min # stars >=3)	Radiation tolerance
Berlin Space Technologies	30/400 arcsec	36 cm ³ / 50g	1 deg/sec	CMOS – low
Blue Canyon Technologies	6/40 arcsec	250 cm ³ / 350 g	2 deg/sec	CMOS – low
Sinclair ST-16	7/70 arcsec	90 cm ³ / 90g	1 deg/sec	CMOS - low
Aerospace OCSD	12/130 arcsec	52 cm ³ / 77g (113g w/2 cams)	1 deg/sec	CMOS – low
Aerospace next gen	6/40 arcsec	52 cm ³ / 77g (113g w/2 cams)	3.9 deg/sec	Silicon on Insulator (SOI) CMOS – better

T Dzamba, J Enright. "Ground Testing Strategies for Verifying the Slew Rate Tolerance of Star Trackers." Sensors 2014, 3939-3964. Enright, D Sinclair, K. C. Fernando. "COTS Detectors for Nanosatellite Star Trackers: A Case Study." Conf. Small Sat. (2011).



Eil Berlin Space Technologies. "ST-200 Flyer: Miniaturised Autonomous Star Tracker" http://www.berlin-space-tech.com/fileadmin/media/BST_ST-200 Flyer.pdf